CODE NO: NR422101 SET NO.1

2005 JAWAHARLAL NEHRU TECHNOLOGICAL UNIVERSITY

IV B.TECH. II SEMESTER SUPPLEMENTARY EXAMINATIONS BOUNDARY LAYER THEORY (AERONAUTICAL ENGINEERING)

JULY -2005

TIME: 3 HOURS MAX MARKS: 80

Answer any FIVE Questions All Questions carry equal marks

1. Estimate the thermal conductivity of helium at $420 \circ C$ and 1 atm. Compare with the measured value of $0.28 \text{ W}/(m \cdot K)$.

2. Write about general of stress state of deformable bodies. Explain the stress Tensor?

3. Consider the plane stagnation-point °ow. Derive the equation for displacement thickness and boundary layer thickness.

4. Develop an implicit numerical algorithm for the two-dimensional unsteady viscous di®usion relation. Comment on a possible solution procedure and possible instability.

5. Derive a relation for skin-friction coe±cient Cf as a function of local Reynolds number Rex for boundary-layer °ow toward a point sink. Compare your result with the Falkner-Skan relations.

6. Assume a boundary-layer velocity pro^-le approximating a Pohlhausen polynomial with any nonzero value of $^$ (have each member of the class select a di $erent^-$). Estimate the critical (instability) value of $Re\pm a$ for this pro ^-le .

7. Consider fully developed turbulent °ow through a duct of square cross section. Taking advantage of the double symmetry, analyze this problem using the log-law, plus a suitable assumption about variation of shear stress around the cross section. Compare your result for ^ with the hydraulic-radius concept.

8. Water at 20oC and 1 atm °ows at 6m /s past a smooth °at plate 1 m long and 60 cm wide. The plate surface temperature is 50oC. Estimate the total heat loss (in W) from one side of the plate.